

NOVEL THERMAL CONTROL CONCEPTS USING MICRO HEAT PIPES - SPACECRAFT THERMAL CONTROL

G. P. “Bud” Peterson

Department of Mechanical Engineering,
Aeronautical Engineering and Mechanics

Rensselaer Polytechnic Institute

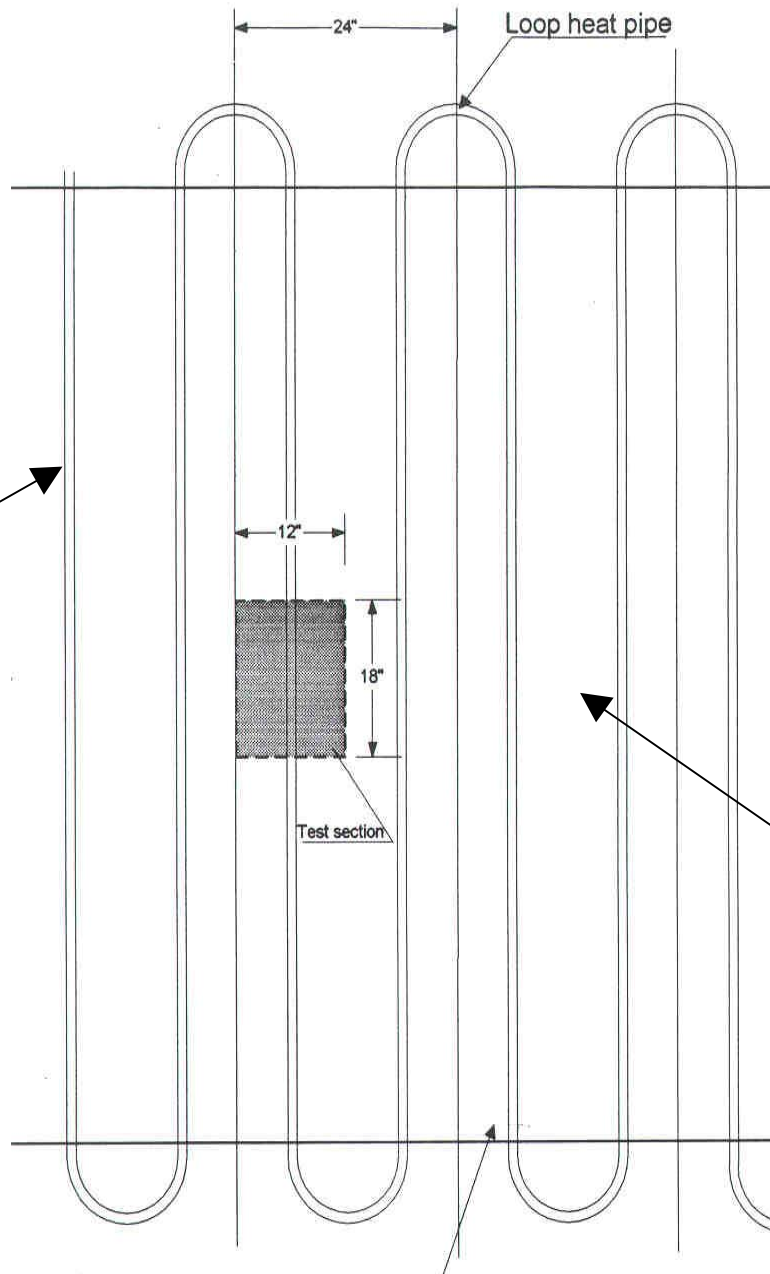
Troy, NY 12180

| | | | | | |
|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------|---------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------|--|
| REPORT DOCUMENTATION PAGE | | | | Form Approved OMB No. 0704-0188 | |
| Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS. | | | | | |
| 1. REPORT DATE (DD-MM-YYYY) 30-05-2001 | | 2. REPORT TYPE Workshop Presentations | | 3. DATES COVERED (FROM - TO) 30-05-2001 to 01-06-2001 | |
| 4. TITLE AND SUBTITLE Novel Thermal Control Concepts Using Micro Heat Pipes - Spacecraft Thermal Control Unclassified | | | | 5a. CONTRACT NUMBER | |
| | | | | 5b. GRANT NUMBER | |
| | | | | 5c. PROGRAM ELEMENT NUMBER | |
| 6. AUTHOR(S) Peterson, G. P. ; | | | | 5d. PROJECT NUMBER | |
| | | | | 5e. TASK NUMBER | |
| | | | | 5f. WORK UNIT NUMBER | |
| 7. PERFORMING ORGANIZATION NAME AND ADDRESS Department of Mechanical Engineering, Aeronautical Engineering and Mechanics Rensselaer Polytechnic Institute Troy, NY12180 | | | | 8. PERFORMING ORGANIZATION REPORT NUMBER | |
| 9. SPONSORING/MONITORING AGENCY NAME AND ADDRESS Office of Naval Research International Field Office Office of Naval Research Washington, DCxxxxx | | | | 10. SPONSOR/MONITOR'S ACRONYM(S) | |
| | | | | 11. SPONSOR/MONITOR'S REPORT NUMBER(S) | |
| 12. DISTRIBUTION/AVAILABILITY STATEMENT APUBLIC RELEASE | | | | | |
| 13. SUPPLEMENTARY NOTES See Also ADM001348, Thermal Materials Workshop 2001, held in Cambridge, UK on May 30-June 1, 2001. Additional papers can be downloaded from: http://www-mech.eng.cam.ac.uk/onr/ | | | | | |
| 14. ABSTRACT ? MICRO HEAT PIPE CONCEPTS ? Wire Bonded Micro Heat Pipe ? Polymer Micro Heat Pipe ? EXPERIMENT FACILITY | | | | | |
| 15. SUBJECT TERMS | | | | | |
| 16. SECURITY CLASSIFICATION OF: | | 17. LIMITATION OF ABSTRACT Public Release | 18. NUMBER OF PAGES 36 | 19. NAME OF RESPONSIBLE PERSON Fenster, Lynn lfenster@dtic.mil | |
| a. REPORT Unclassified | b. ABSTRACT Unclassified | c. THIS PAGE Unclassified | 19b. TELEPHONE NUMBER International Area Code Area Code Telephone Number 703767-9007 DSN 427-9007 | | |
| | | | | Standard Form 298 (Rev. 8-98) Prescribed by ANSI Std Z39.18 | |

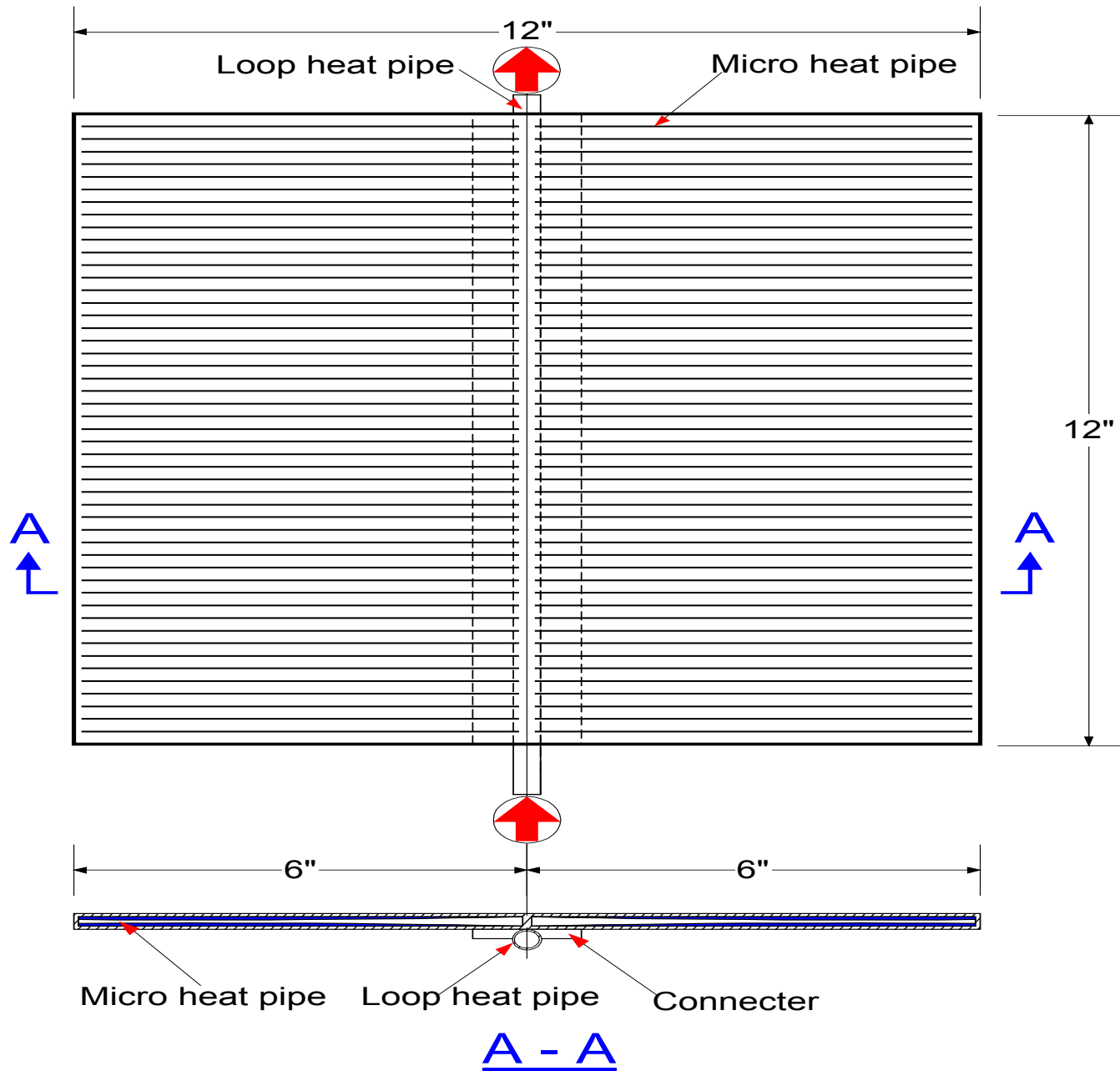
Introduction

- **BACKGROUND**
- **MICRO HEAT PIPE CONCEPTS**
 - **Wire Bonded Micro Heat Pipe**
 - **Polymer Micro Heat Pipe**
- **EXPERIMENT FACILITY**
- **RESULTS & DISCUSSION**
- **CONCLUSIONS**

Loop heat
pipe coils



Space occupied
by flexible fins

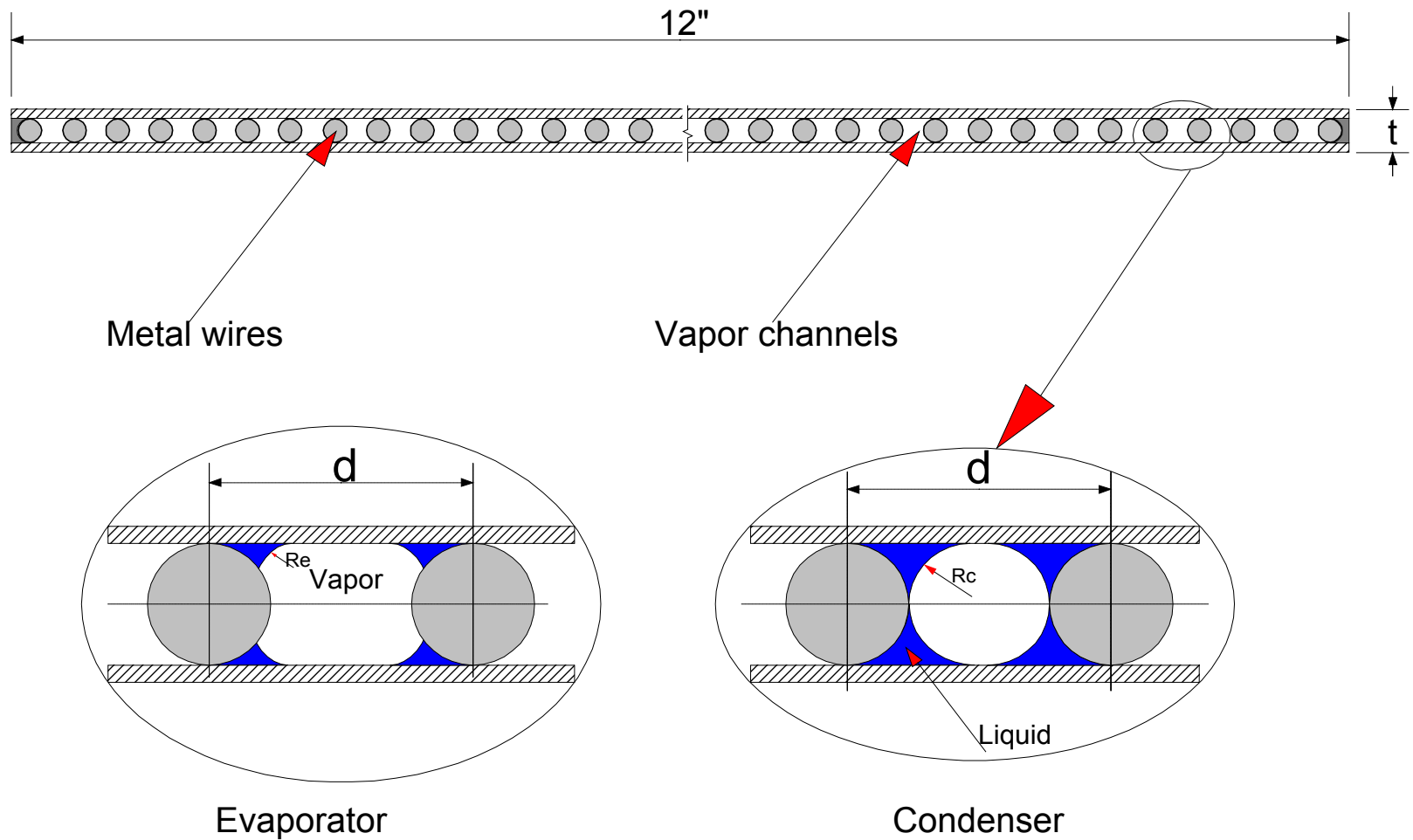


Initial Concepts

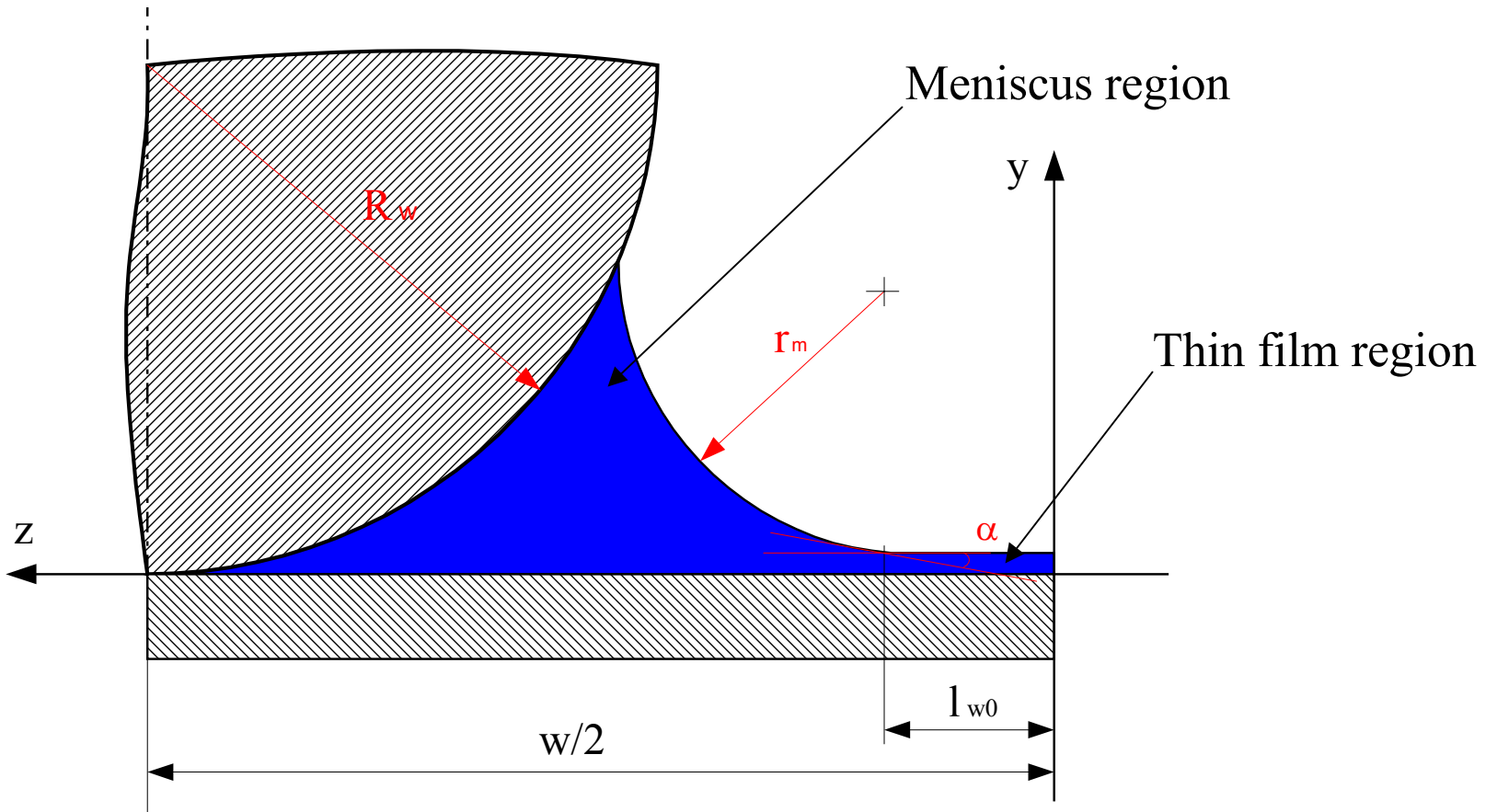
From an initial group of six concepts, the following were selected for further study

- **Wire bonded heat pipe array**
- **Flexible Polymer Heat Pipe**

Flexible Wire Bonded Micro Heat Pipe



Liquid Distribution



Geometric Parameters

$$\beta_1 + \beta_2 + \alpha = \frac{\pi}{2} \quad R_w \sin^2 \beta_1 = r_m \cos \beta_1 \sin \beta_2$$

$$\beta_1 = a \tan \left[\frac{1}{2R_w} \left(-r_m \sin \alpha + \left((r_m \sin \alpha)^2 + 4R_w r_m \cos \alpha \right)^{\frac{1}{2}} \right) \right]$$

$$A_{c,l} = 2R_w r_m \sin \beta_1 \sin \beta_2 - R_w^2 (\beta_1 - \sin \beta_1 \cos \beta_2) - r_m^2 (\beta_2 - \sin \beta_2 \cos \beta_2)$$

$$A_{c,v} = R_w (2w - \pi R_w) - 4A_{c,l}$$

$$l_{ac} = 2R_w \beta_1$$

$$l_{ab} = \left[(r_m \sin \alpha)^2 + 4R_w r_m \cos \alpha \right]^{\frac{1}{2}} - r_m \sin \alpha$$

$$l_{cb} = 2r_m \beta_2$$

Fluid Flow Model

Continuity Equations

$$\frac{d(u_l A_{c,l})}{dx} - v_{l,i} p_i = 0$$

$$\frac{d(u_v A_{c,v})}{dx} - v_{v,i} p_i = 0$$

Momentum Equations

$$-\rho_l \left(2A_{c,l} u_l \frac{du_l}{dx} + u_l^2 \frac{dA_{c,l}}{dx} \right) - A_{c,l} \frac{dP_l}{dx} + p_{l,i} \tau_{l,i} + p_{l,w} \tau_{l,w} - g \rho_l A_{c,l} \sin \theta = 0$$

$$\rho_v \left(2A_{c,v} u_v \frac{du_v}{dx} + u_v^2 \frac{dA_{c,v}}{dx} \right) + A_{c,v} \frac{dP_v}{dx} + p_{v,i} \tau_{v,i} + p_{v,w} \tau_{v,w} + g \rho_v A_{c,v} \sin \theta = 0$$

Energy Equations

$$v_{l,i} = \frac{q'' w}{2 \rho_l p_{l,i} h'_{fg}}$$

$$v_{v,i} = \frac{2 w q''}{\rho_v p_{v,i} h'_{fg}}$$

Fluid Flow Model

The condenser exposed in radiation:

$$q_c'' = h_{c,o} (T_v - T_{\text{sink}}) \left(1 + \frac{h_{c,o}}{h_c} \right)^{-1}$$

$$h_{c,o} = \varepsilon \sigma_0 (T_{w,o} + T_{\text{sink}}) (T_{w,o}^2 + T_{\text{sink}}^2)$$

$$v_{v,i} = \frac{2wh_{c,o} (T_v - T_{\text{sink}}) \left(1 + \frac{h_{c,o}}{h_c} \right)^{-1}}{\rho_v p_{v,i} h_{fg}}$$

In the adiabatic section:

$$q_a'' = 0$$

In the evaporator section:

$$q_e'' = \text{constant}$$

$$v_{l,i} = \frac{wh_{c,o} (T_v - T_{\text{sink}}) \left(1 + \frac{h_{c,o}}{h_c} \right)^{-1}}{2\rho_l p_{l,i} h_{fg}}$$

Heat Transfer in the Evaporator

Heat transfer in evaporating film:

$$\Delta Q_1 = 0$$

$$\Delta Q_{2+3} = \int_{z'_2}^{z'_2} \frac{T_{w,i} - T_v}{\frac{1}{h_{ph}} + \frac{\delta(z')}{k_l}} dz'$$

$$\Delta Q_{2+3} = \int_{z'_2}^{z'_2} \frac{T_{w,i} - T_v}{\frac{1}{3.2 \rho_v h_{fg} \left(\frac{T_v}{R_g} \right)^{0.5}} + \frac{\left(\delta_0 - r_m + (r_m^2 + z'^2)^{0.5} \right)}{k_l}} dz'$$

$$\Delta Q_4 = \int_{z'_2}^{2r_m \sin \beta_1 \sin \beta_2} \frac{(T_{w,i} - T_v)}{\left(1/h_{ph} + z'^2 / 2r_m k_l \right)} dz'$$

$$\bar{h}_{ph} = 3.2 \rho_v h_{fg} \sqrt{\frac{R_g}{T_v}}$$

$$= 1.41 (T_{w,i} - T_v) (h_{pc} k_l r_m)^{\frac{1}{2}} \left[\frac{\pi}{2} - a \tan \left(\frac{h_{ph} \delta_2}{k_l} \right)^{\frac{1}{2}} \right]$$

$$h_{e,men} = \frac{\Delta Q_{2+3} + \Delta Q_4}{(T_{w,i} - T_v) l_{ab}}$$

$$\bar{h}_e = \left\{ \left[\frac{2l_{ab}}{w} h_{e,men} \right]^{-1} + \frac{t_w}{k_w} \right\}^{-1}$$

$$T_{w,o} = T_{sink} + \frac{T_v - T_{sink}}{h_{o,e}} \left(\frac{1}{h_{o,e}} + \frac{1}{\bar{h}_{i,e}} \right)^{-1}$$

Heat Transfer in the Condenser

The thickness of the condensed liquid film:

$$-\frac{\sigma \delta^3}{3\nu_l} \frac{d^3 \delta}{dz^3} = \frac{k_l (T_{sat} - T_w)}{h_{fg}} \int \frac{1}{\delta} dz \quad \text{B.C:} \quad \left. \frac{d\delta}{dz} \right|_{z=0} = 0 \quad \left. \frac{d^3 \delta}{dz^3} \right|_{z=0} = 0$$

Assuming the thickness of liquid film: $\left. \frac{d^2 \delta}{dz^2} \right|_{z=l_{w0}} = \frac{1}{r_m} \quad \left. \frac{d\delta}{dz} \right|_{z=l_{w0}} = \tan \alpha$

$$\frac{\delta}{w} = C_0 + C_1 \frac{z}{w} + C_2 \frac{z^2}{w^2} + C_3 \frac{z^3}{w^3} + C_4 \frac{z^4}{w^4}$$

$$\bar{h}_{c,f} = \frac{k_l}{\delta} \quad \bar{h}_{c,f} = \frac{k_l}{w C_0} = \left[\frac{h_{fg} \rho_l \sigma k_l^3 (l_{w0} / r_m - \tan \alpha)}{\mu_l l_{w0}^3 (T_v - T_w)} \right]^{\frac{1}{4}}$$

Heat Transfer in the Condenser

Film thickness of the liquid meniscus:

$$\delta = \delta \Big|_{s=l_{w0}} + r_m \cos \alpha - \left[(r_m \cos \alpha)^2 - (z - l_{w0})^2 - 2r_m (z - l_{w0}) \sin \alpha \right]^{\frac{1}{2}}$$

Heat transfer coefficient in meniscus:

$$\bar{h}_{c,men} = \int_{w0}^{2r_m \sin \beta_1 \sin \beta_2} \frac{k_l}{l_{ab} \left\{ \delta \Big|_{s=l_{w0}} + r_m \cos \alpha - \left[(r_m \cos \alpha)^2 - (z - l_{w0})^2 - 2r_m (z - l_{w0}) \sin \alpha \right]^{0.5} \right\}} dz$$

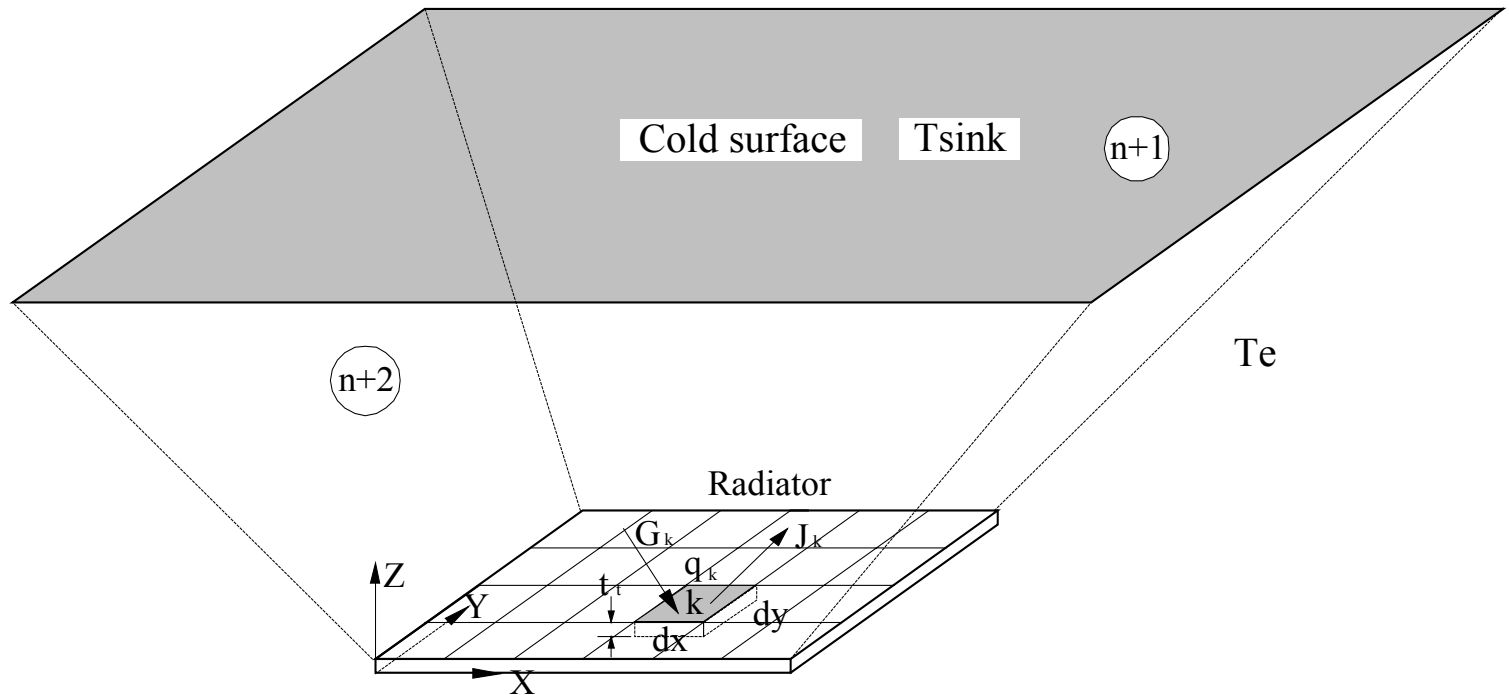
Heat transfer coefficient in the condenser

$$\bar{h}_c = \left\{ \left[\frac{2l_{w0}}{w} \bar{h}_{c,f} + \left(1 - \frac{2l_{w0}}{w} \right) \bar{h}_{c,men} \right]^{-1} + \frac{t_w}{k_w} \right\}^{-1}$$

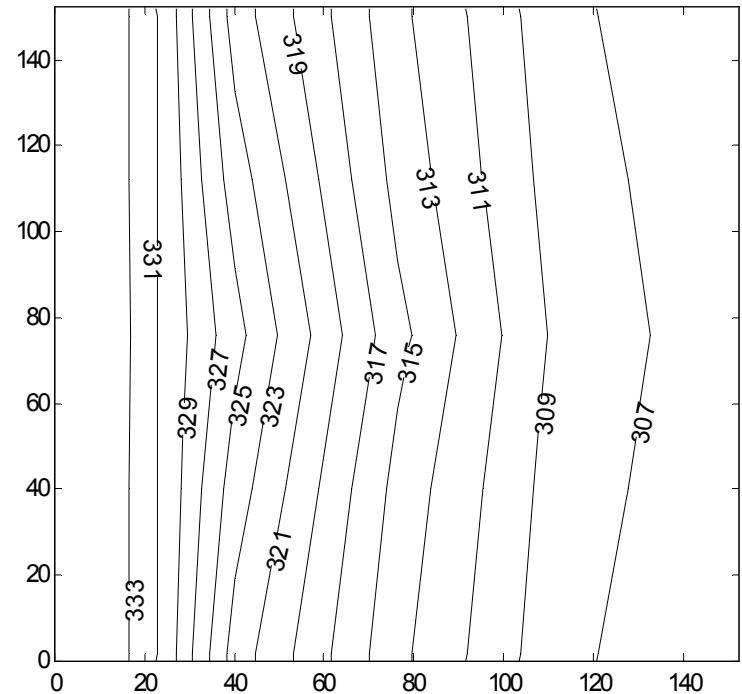
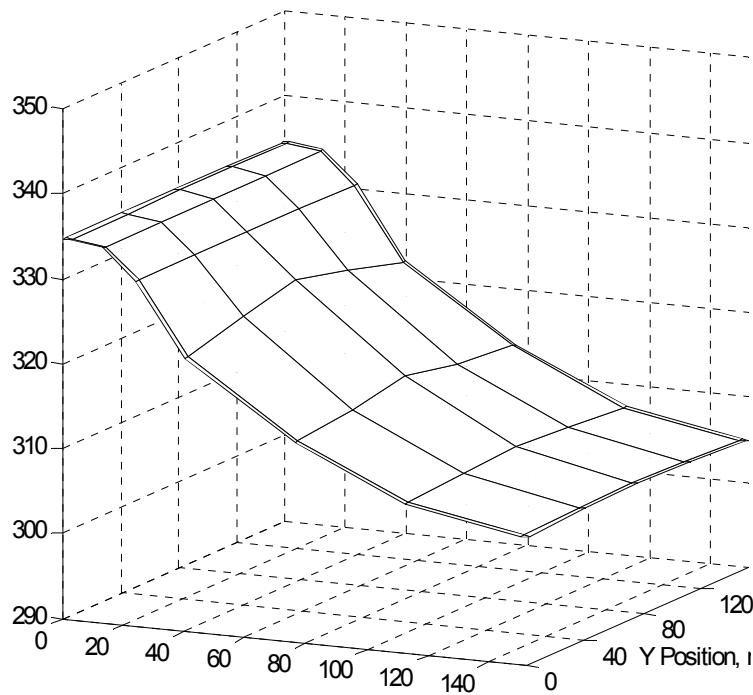
$$T_{w,o} = T_{c,o} + (T_v - T_{c,o}) \frac{1}{h_{c,o}} \left(\frac{1}{h_{c,o}} + \frac{1}{\bar{h}_c} \right)^{-1}$$

$$h_{c,o} = \varepsilon \sigma_0 (T_{w,o} + T_{\sin k}) (T_{w,o}^2 + T_{\sin k}^2)$$

Radiation Heat Transfer

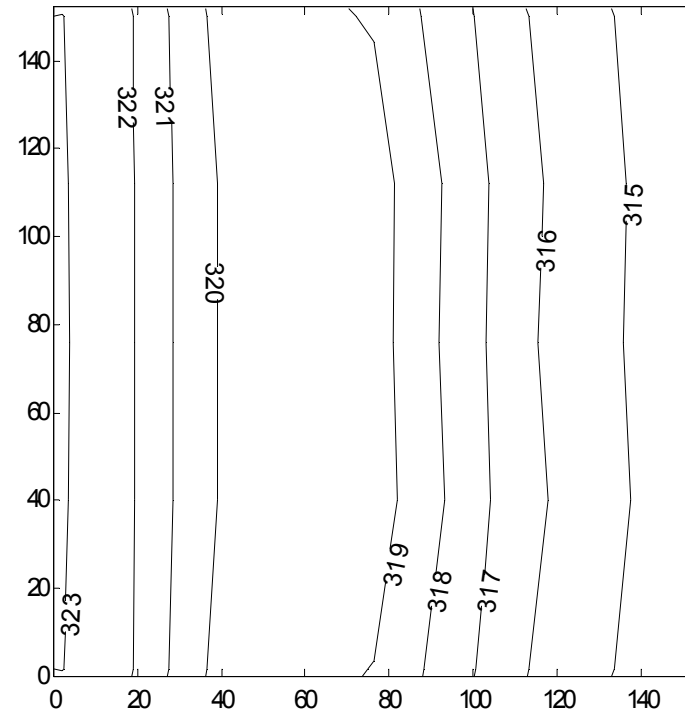
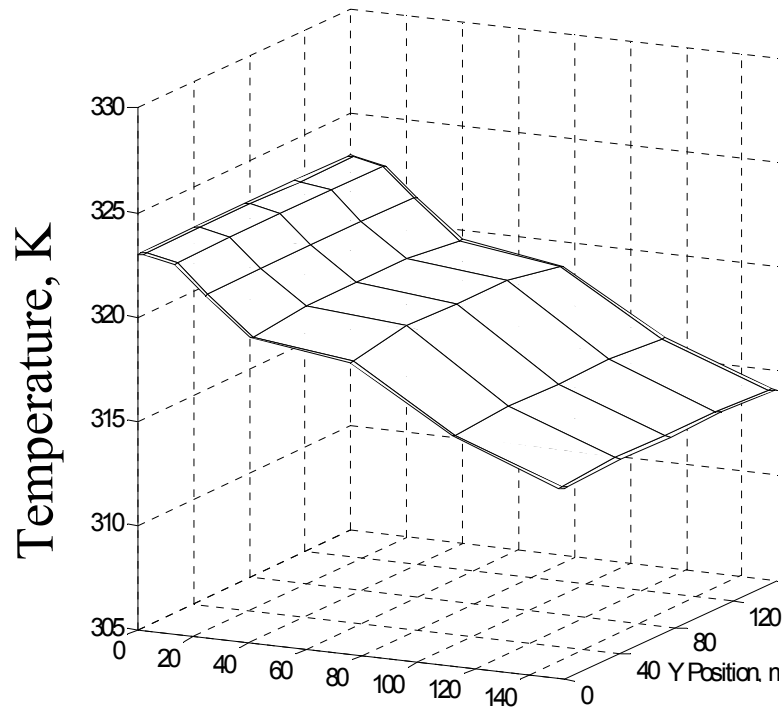


Temperature Distributions on the Wire Bonded Micro Heat Pipe Radiator

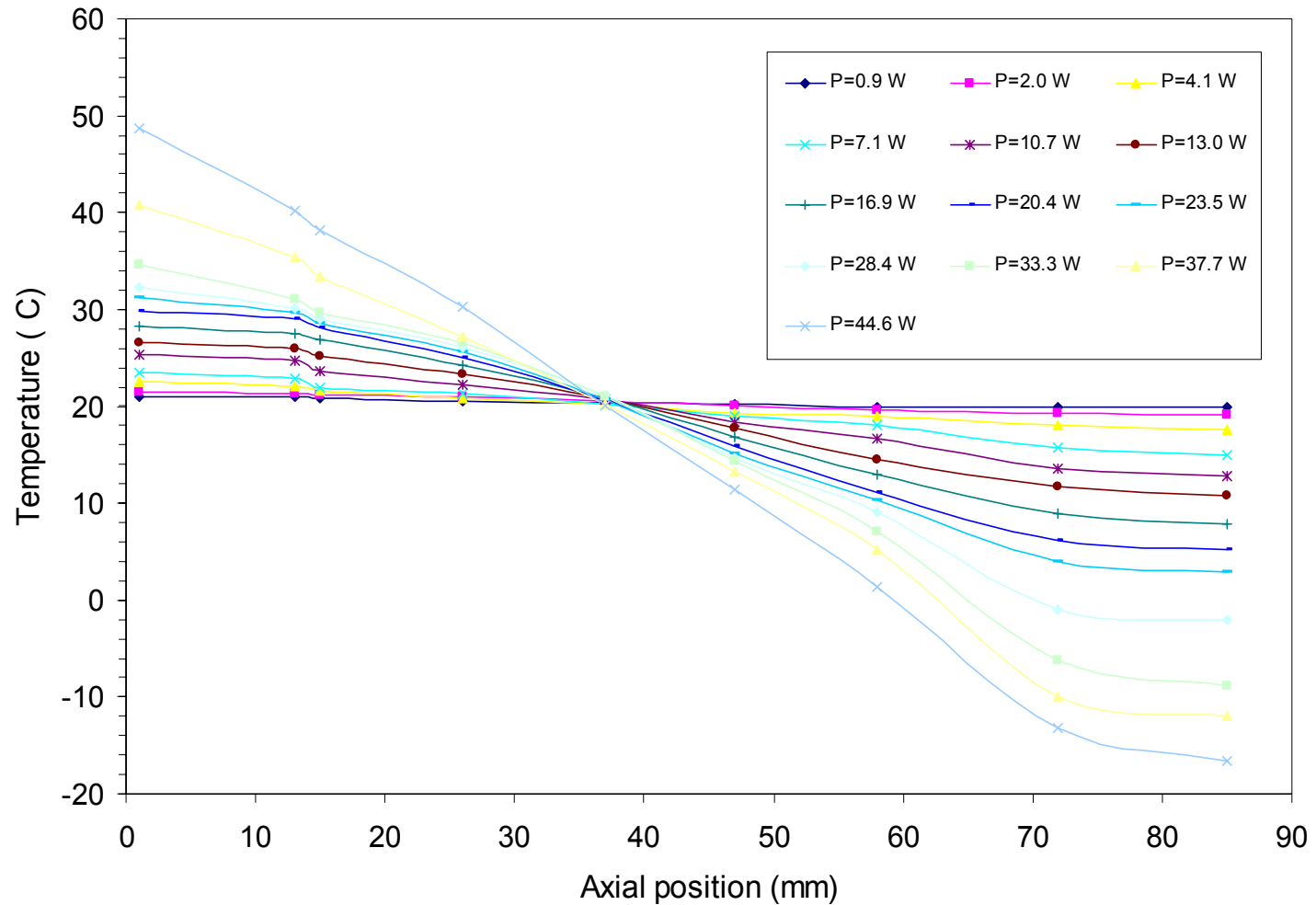


No fluid charge, $d_w=0.813$ mm

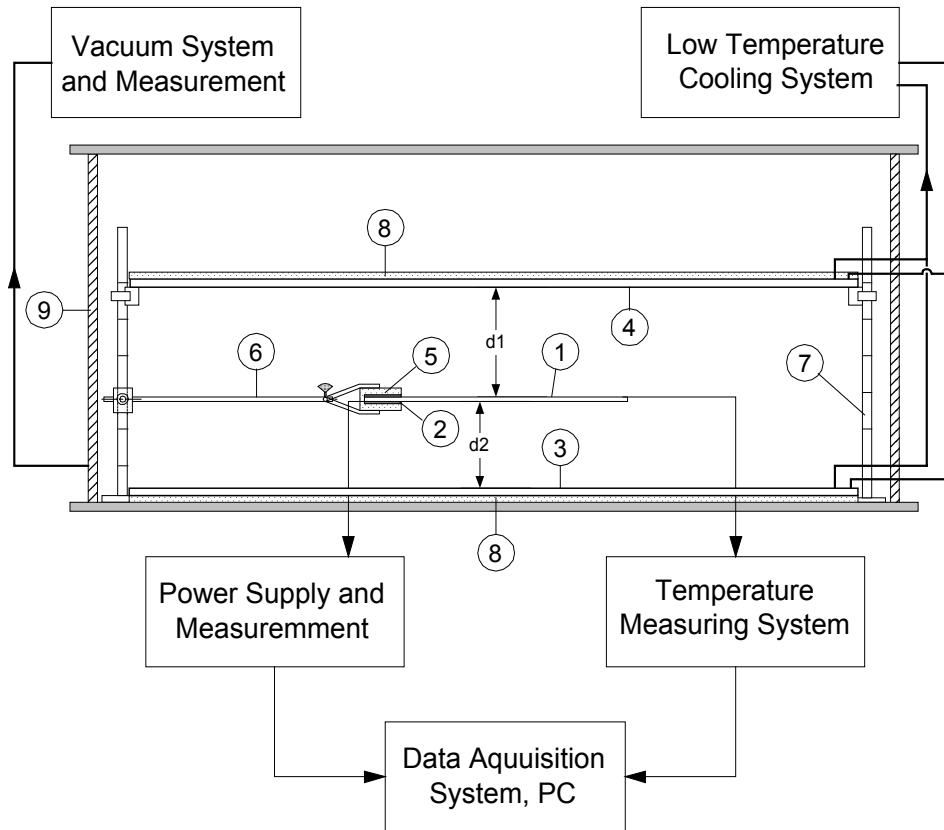
Temperature Distributions on the Wire Bonded Micro Heat Pipe Radiator



Charged with optimum charge, $d_w=0.813$ mm



Test Facility for Radiation Environment



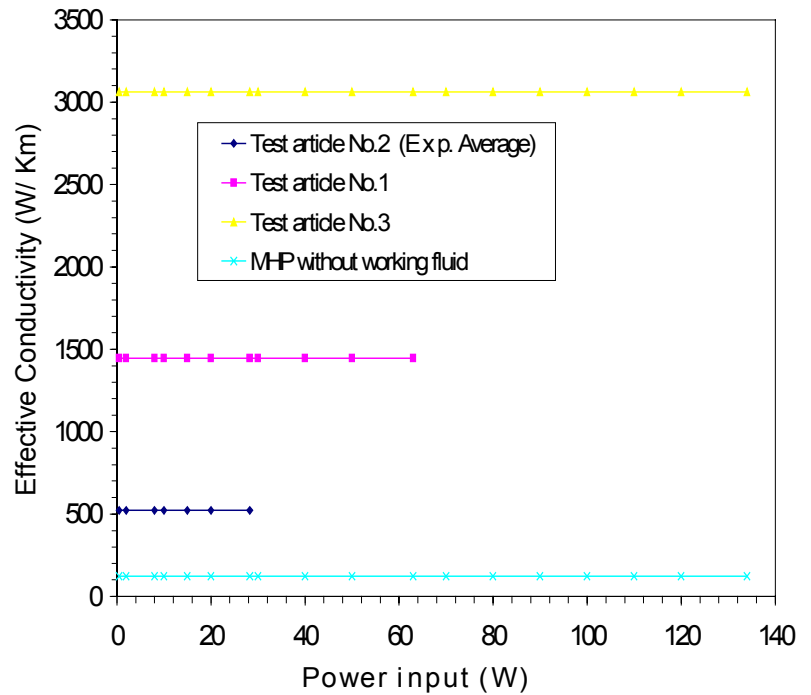
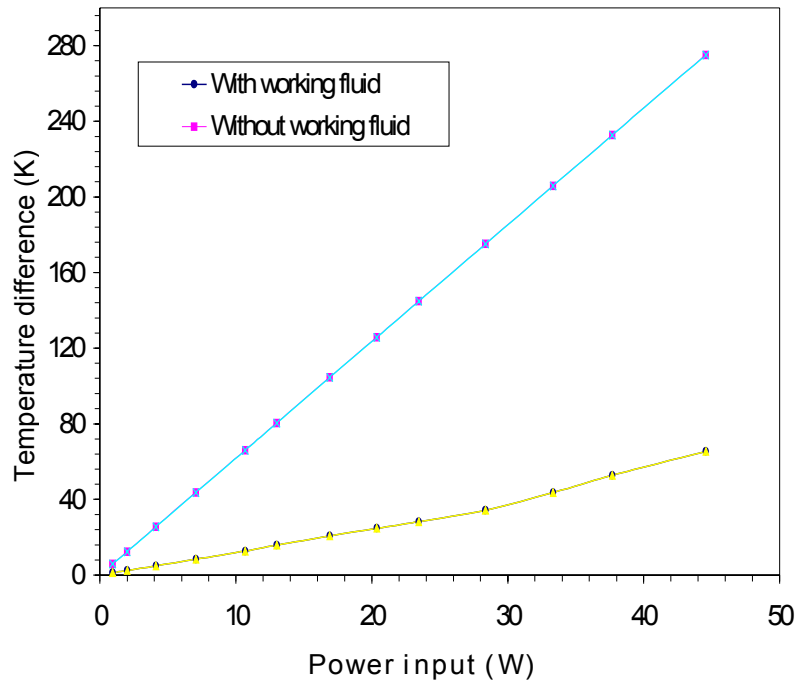
1. Micro heat pipe radiator
2. Electric heater
3. Bottom cold plate
4. Top cold plate
5. Electric heater insulation
6. Adjustable support level
7. Adjustable support feet
8. Insulation material
9. Vacuum chamber

Test Articles

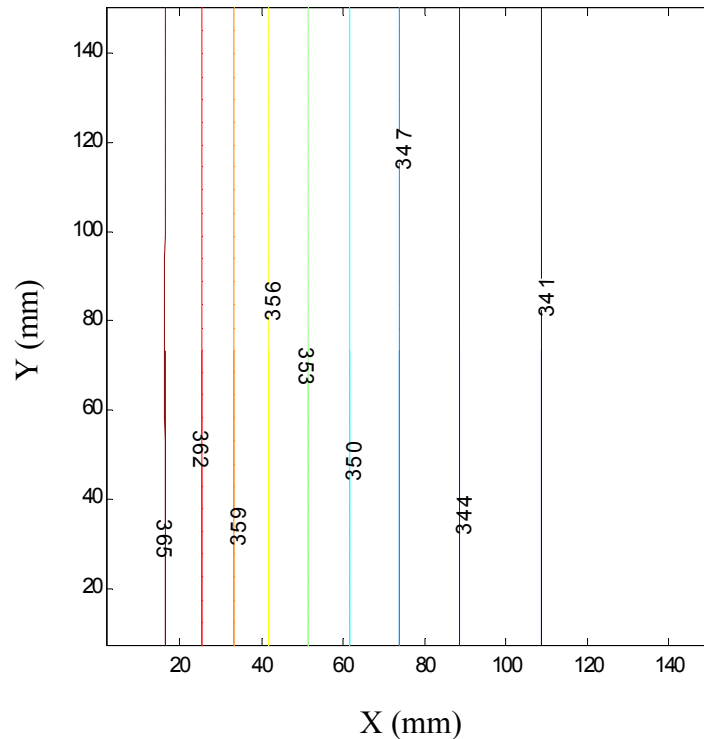
| Prototype | No. 1 | No. 2 | No. 3 |
|----------------------------|-----------|-----------|-----------|
| Material | Aluminum | Aluminum | Aluminum |
| Working fluid | Acetone | Acetone | Acetone |
| Total Dimension (mm) | 152×152.4 | 152×152.4 | 152×152.4 |
| Thickness of sheet (mm) | 0.40 | 0.40 | 0.40 |
| Diameter of wire (mm) | 0.50 | 0.80 | 0.50 |
| Number of wires | 43 | 43 | 95 |

Experimental Results

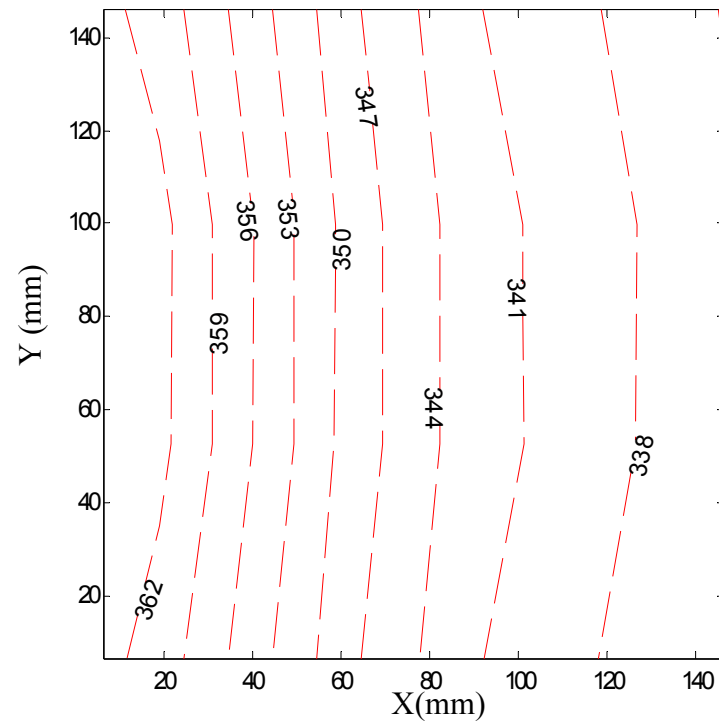
• Effective Thermal Conductivity



Comparison of Experimental and Predicted Results

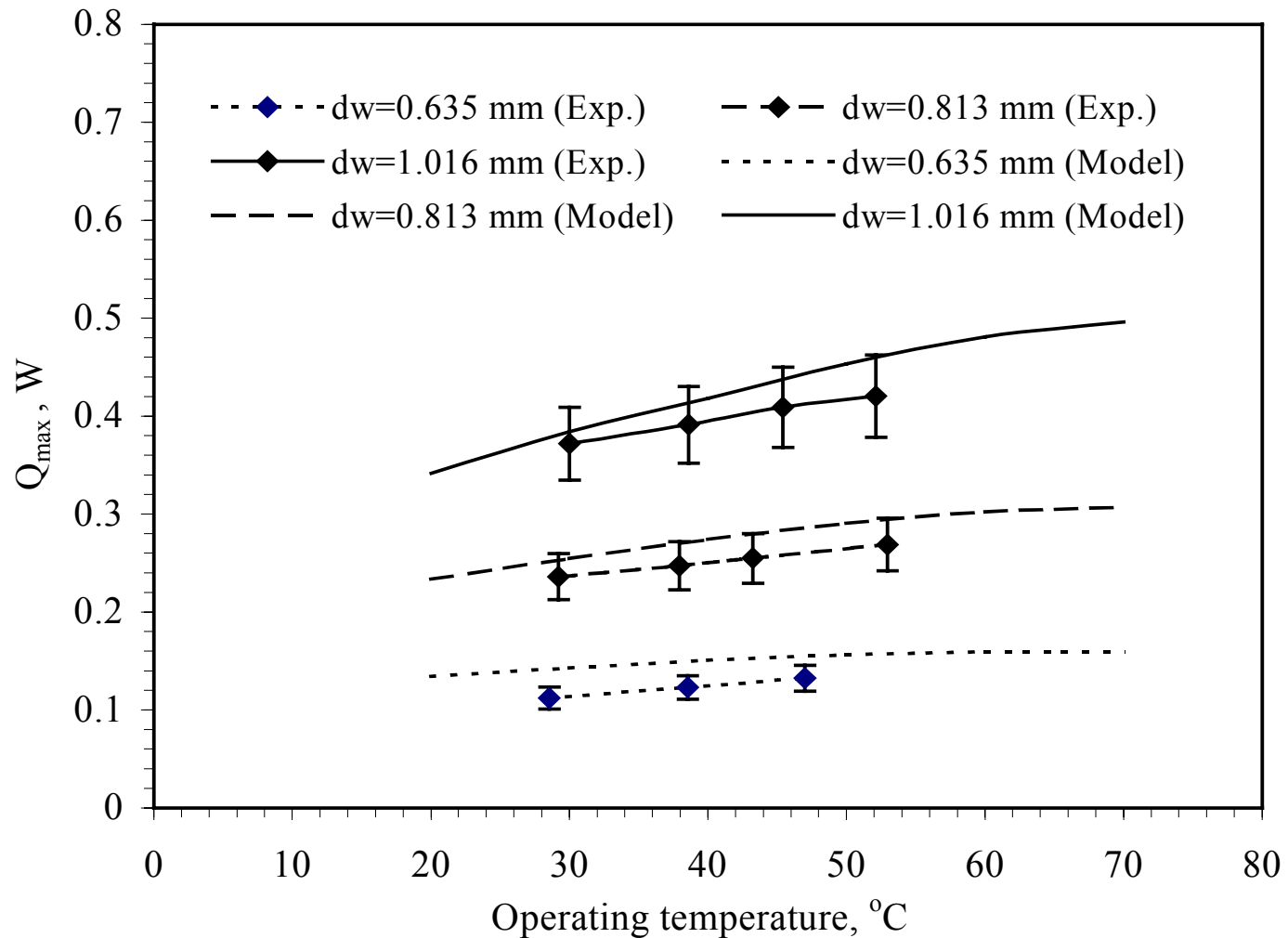


Modeled Temperature Distribution

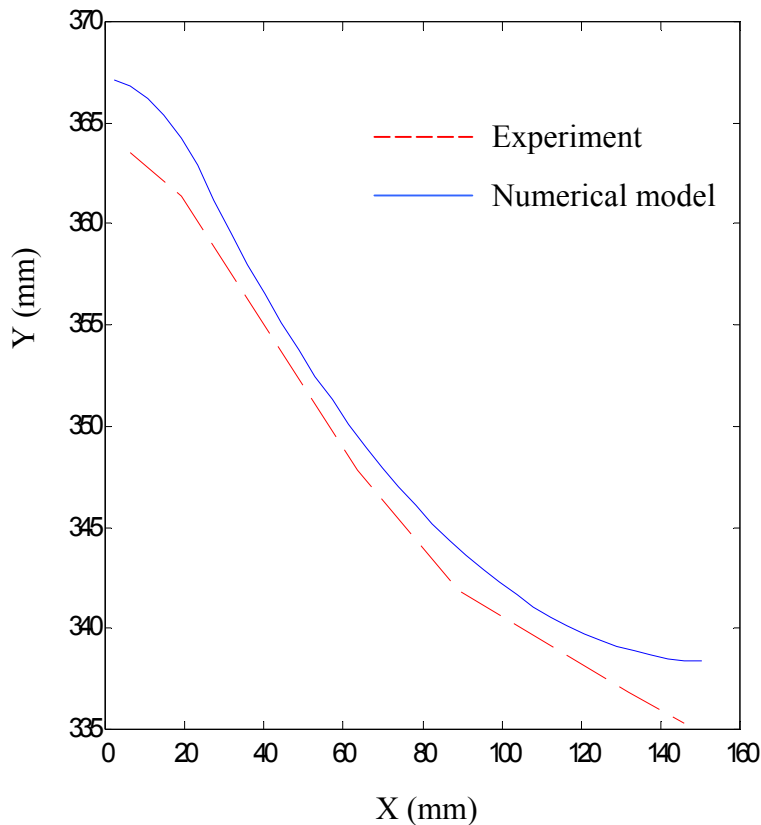


Measured Temperature Distribution

Comparison of the Predicted Experimental Results

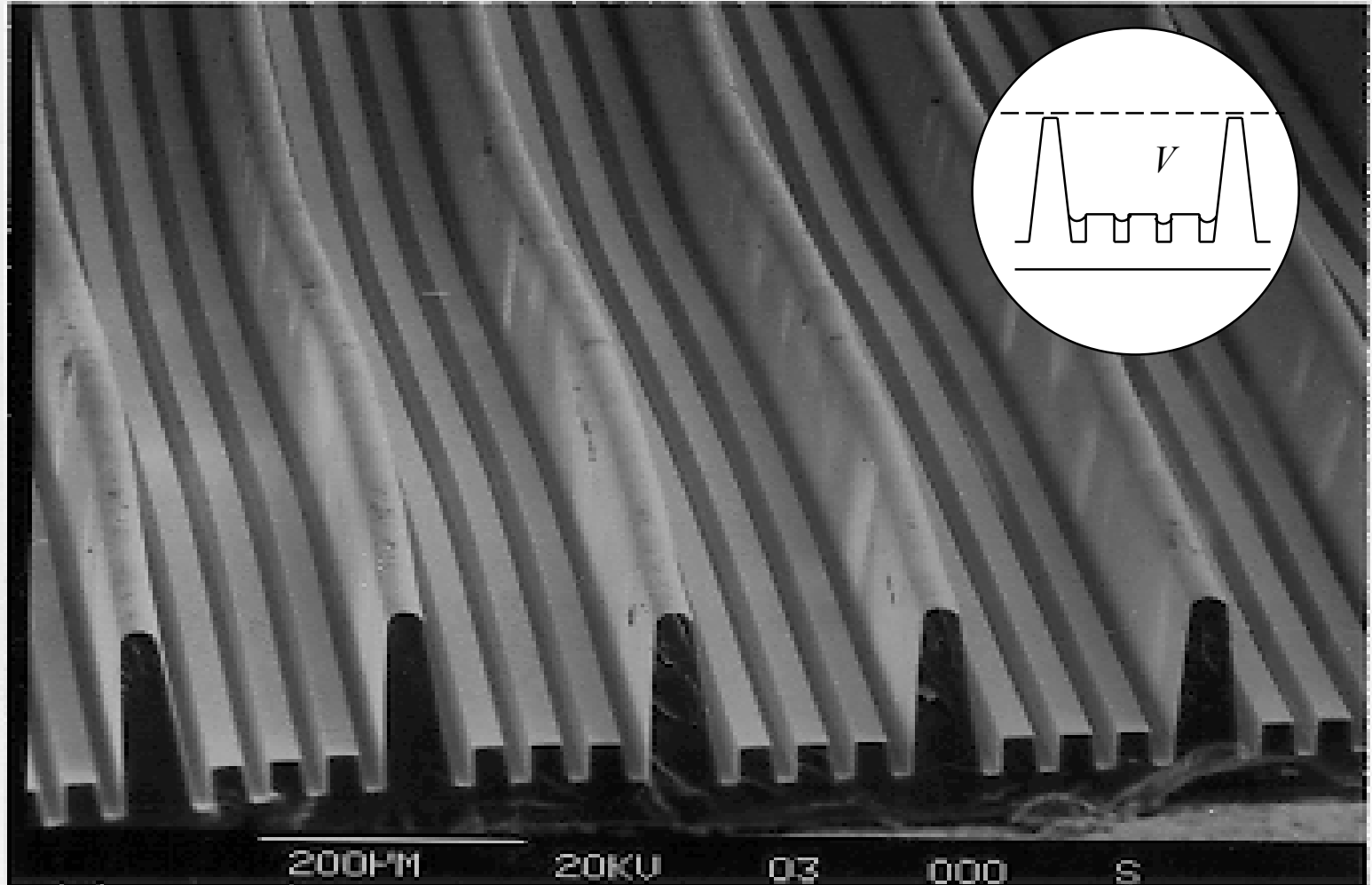


Comparison of Experimental and Predicted Results

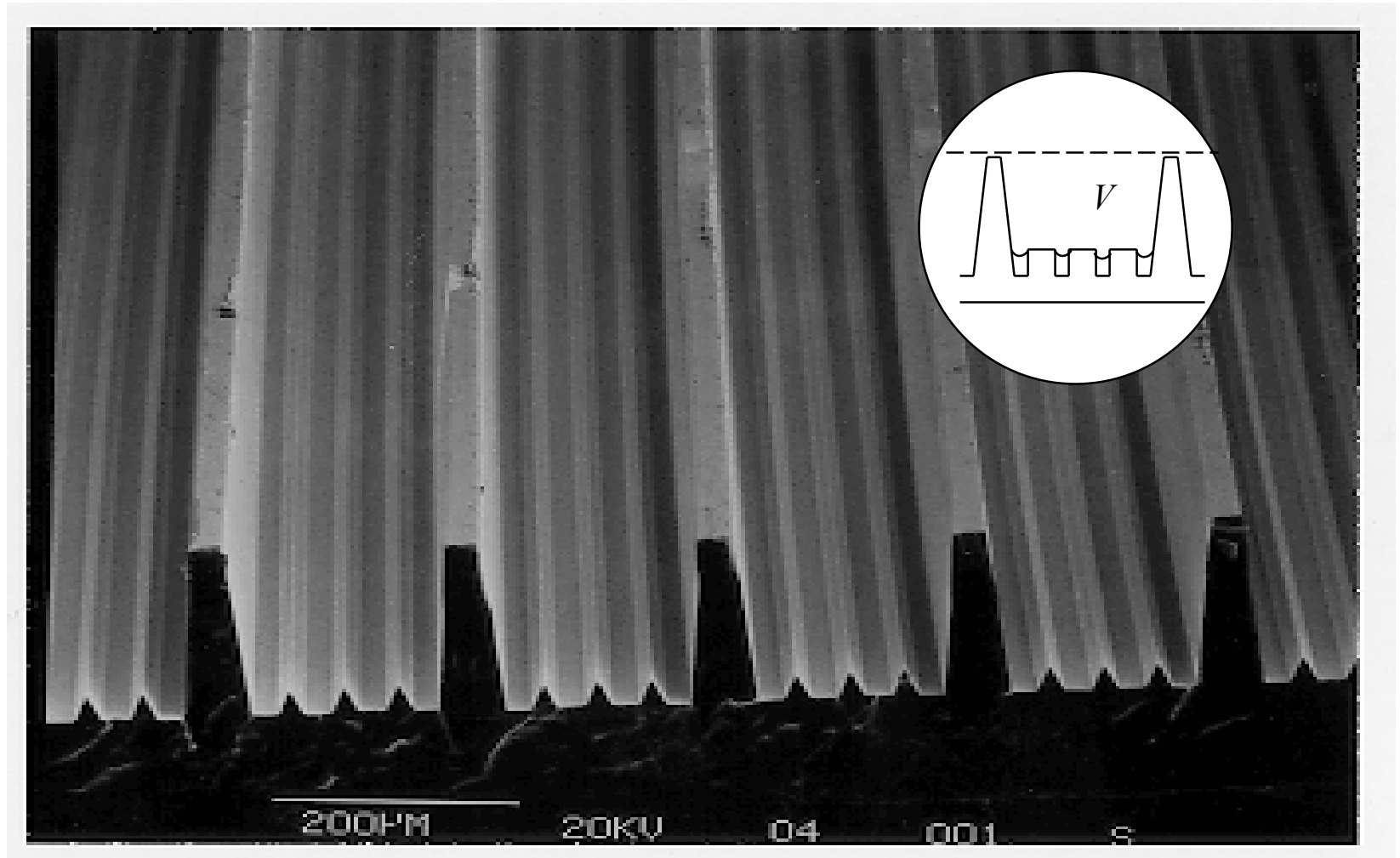


- Temperature distribution obtained from the experiment was slightly lower than from the model.
- Heat loss may result from conduction through the support and insulation and edge radiation
- Total difference is less than 6.8%

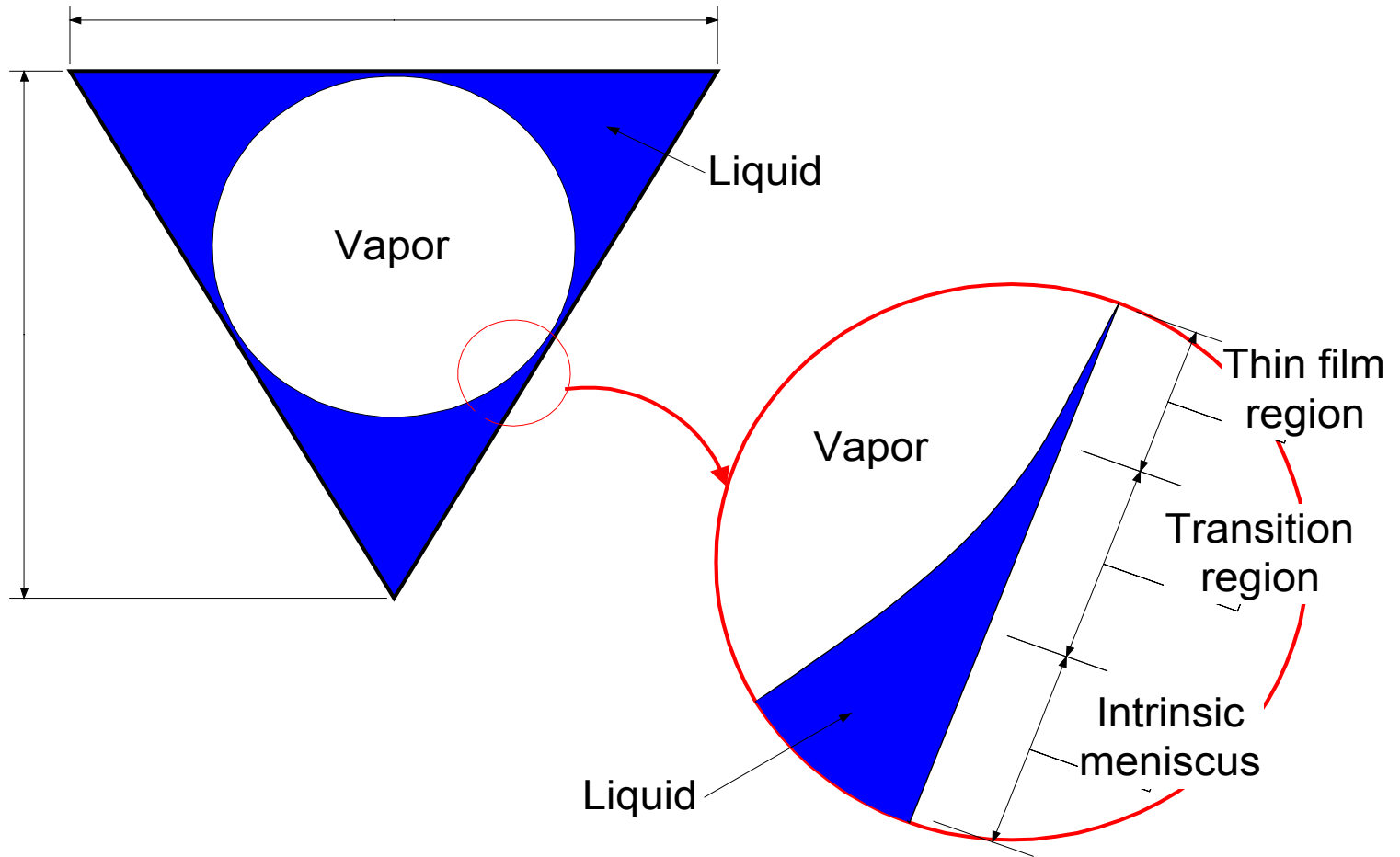
Rectangular Polymer Heat Pipe



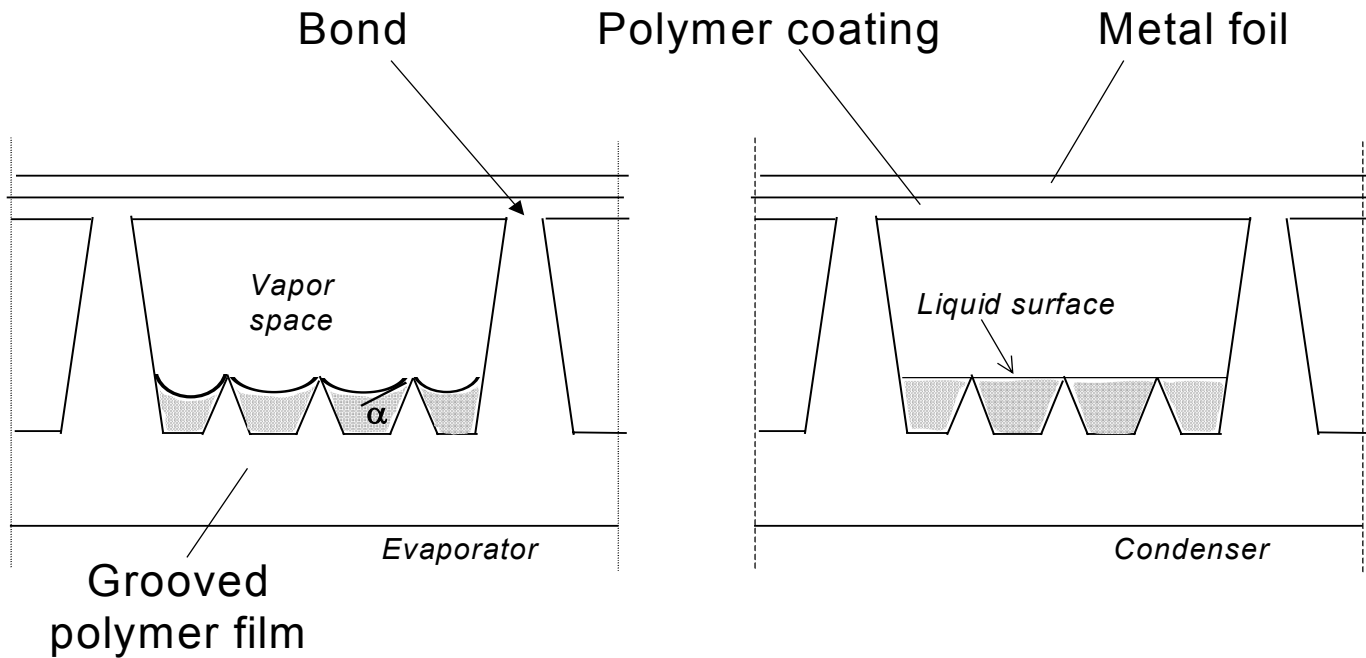
Triangular Polymer Heat Pipe



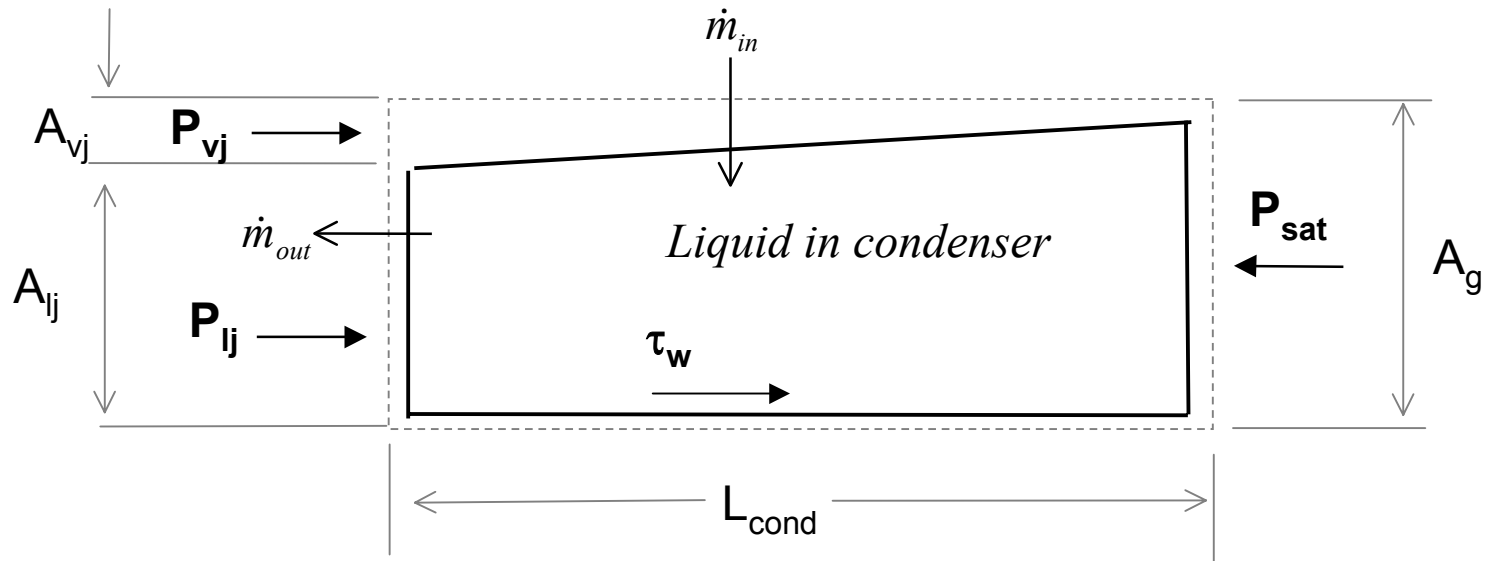
Interfacial Region



Polymer Micro Heat Pipe



Momentum Balance



(Fig. 13)

Numerical Model

The governing equation for two-dimensional plate conduction combined with a radiation boundary condition can be expressed as,

$$\frac{\partial}{\partial x}(ttk_{eff,x} \frac{\partial T}{\partial x}) + \frac{\partial}{\partial y}(ttk_{eff,y} \frac{\partial T}{\partial y}) - q_k + q_0 = 0 \quad (1)$$

if q_k is the net radiation transfer through the control volume k , this equation can be expressed as,

$$\sum_{j=1}^{n+2} \left(\frac{\delta_{kj}}{\varepsilon_j} - F_{k-j} \frac{1 - \varepsilon_j}{\varepsilon_j} \right) q_j = \sum_{j=1}^{n+2} (\delta_{kj} - F_{k-j}) \sigma T_j^4 \quad (2)$$

where δ_{kj} is the Kronecker delta defined as,

$$\delta_{kj} = \begin{cases} 1 & \text{when } k=j \\ 0 & \text{when } k \neq j \end{cases}$$

Numerical Model

For the case of constant thermal effective conductivity for the micro heat pipe array, Eq. (1) becomes,

$$K_{eff,x} \frac{\partial^2 T}{\partial x^2} + K_{eff,y} \frac{\partial^2 T}{\partial y^2} - q_k + q_0 = 0 \quad (3)$$

where $K_{eff,x} = t k_{eff,x}$, and $K_{eff,y} = t k_{eff,y}$

The discretization equation is,

$$a_1 T_{i-1,j}^n - a_2 T_{i,j}^n + a_1 T_{i+1,j}^n = c_1 - c_2 (T_{i,j}^{n-1} + T_{i,j-1}^{n-1}) \quad (4)$$

where

$$a_1 = \frac{K_{eff,x}}{\Delta x^2}, \quad a_2 = 2 \left(\frac{K_{eff,x}}{\Delta x^2} + \frac{K_{eff,y}}{\Delta y^2} \right)$$
$$c_1 = (q_k - q_0), \quad c_2 = \frac{K_{eff,y}}{\Delta y^2}$$

Numerical Model

Adiabatic boundary condition on the edges are assumed , i.e.,

$$\frac{\partial T}{\partial x} = 0 \quad \left\{ \begin{array}{ll} \text{at } x = 0, & 0 \leq y \leq 152.4 \text{ mm} \\ \text{at } x = 152 \text{ mm}, & 0 \leq y \leq 152.4 \text{ mm} \end{array} \right.$$

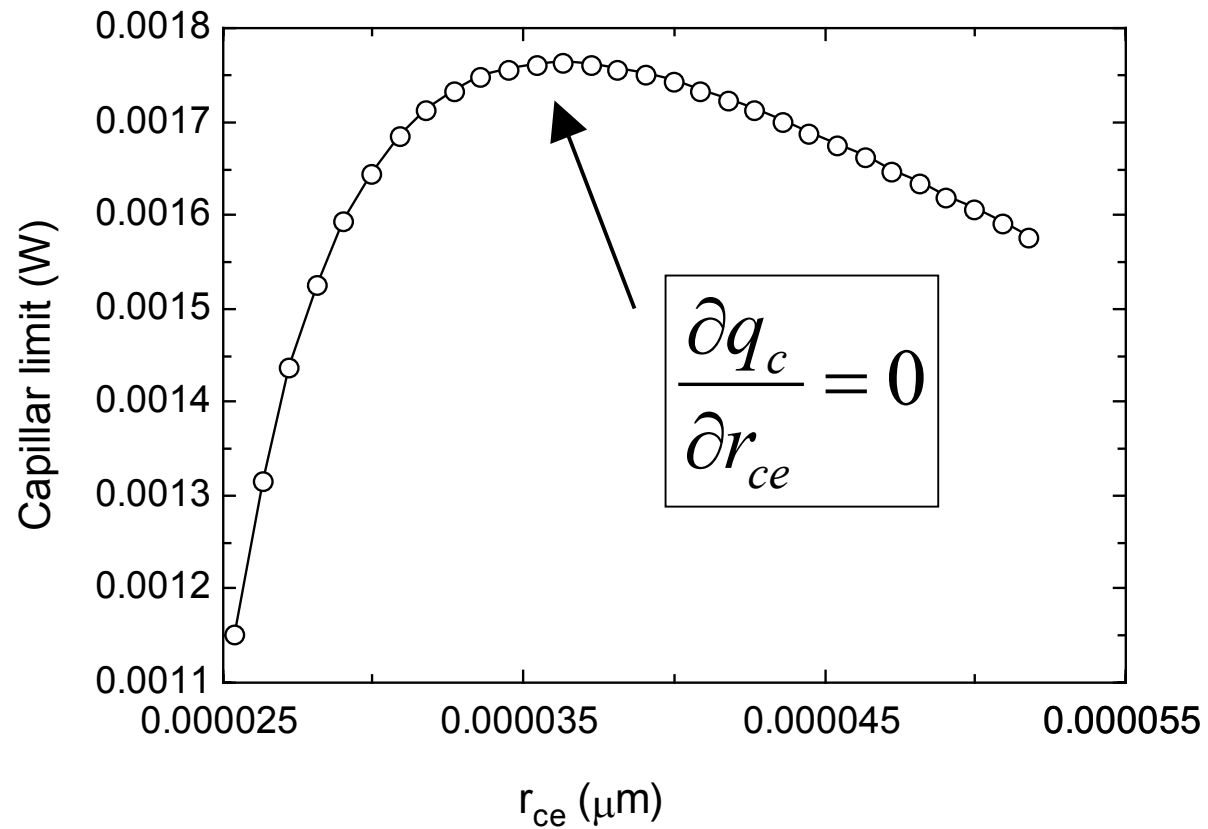
$$\frac{\partial T}{\partial y} = 0 \quad \left\{ \begin{array}{ll} \text{at } y = 0, & 0 \leq x \leq 152 \text{ mm} \\ \text{at } y = 152.4 \text{ mm}, & 0 \leq x \leq 152 \text{ mm} \end{array} \right.$$

and

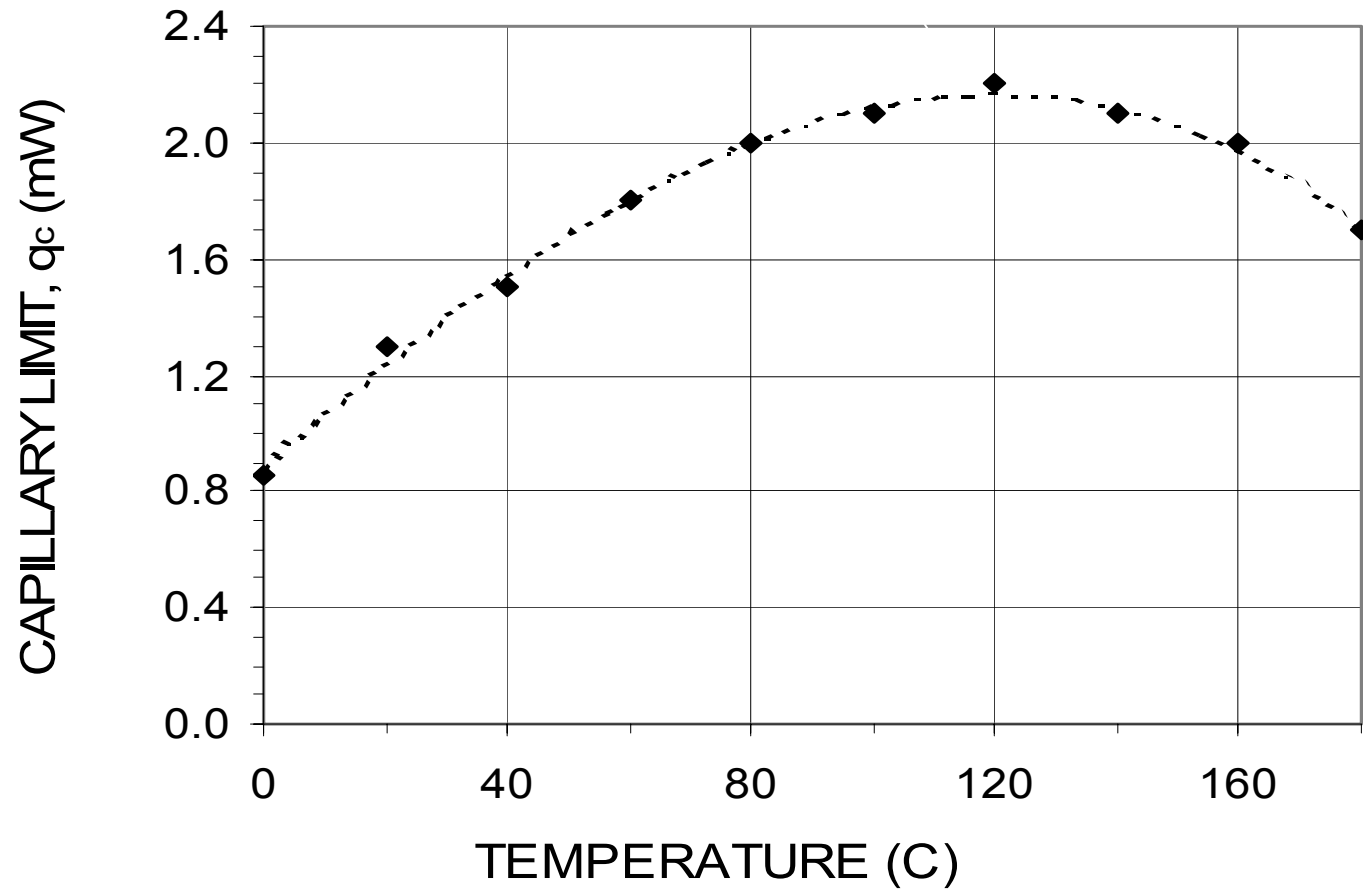
$$q_0 = q_{\text{input}}, \quad 0 \leq x \leq 25.4 \text{ mm}, \quad 0 \leq y \leq 152.4 \text{ mm}$$

$$q_0 = 0, \quad x > 25.4 \text{ mm}, \quad 0 \leq y \leq 152.4 \text{ mm}$$

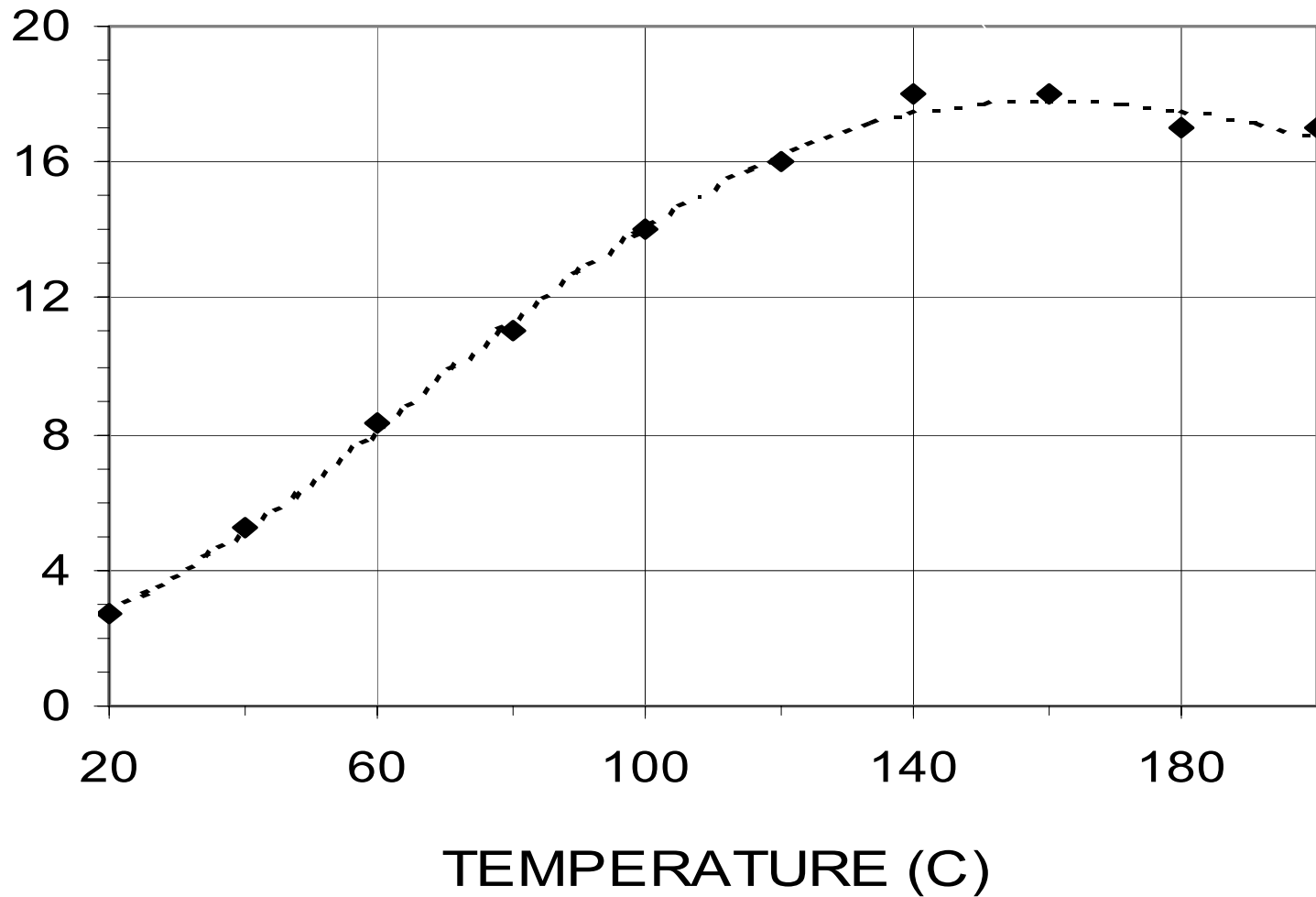
Optimum Capillary Radius



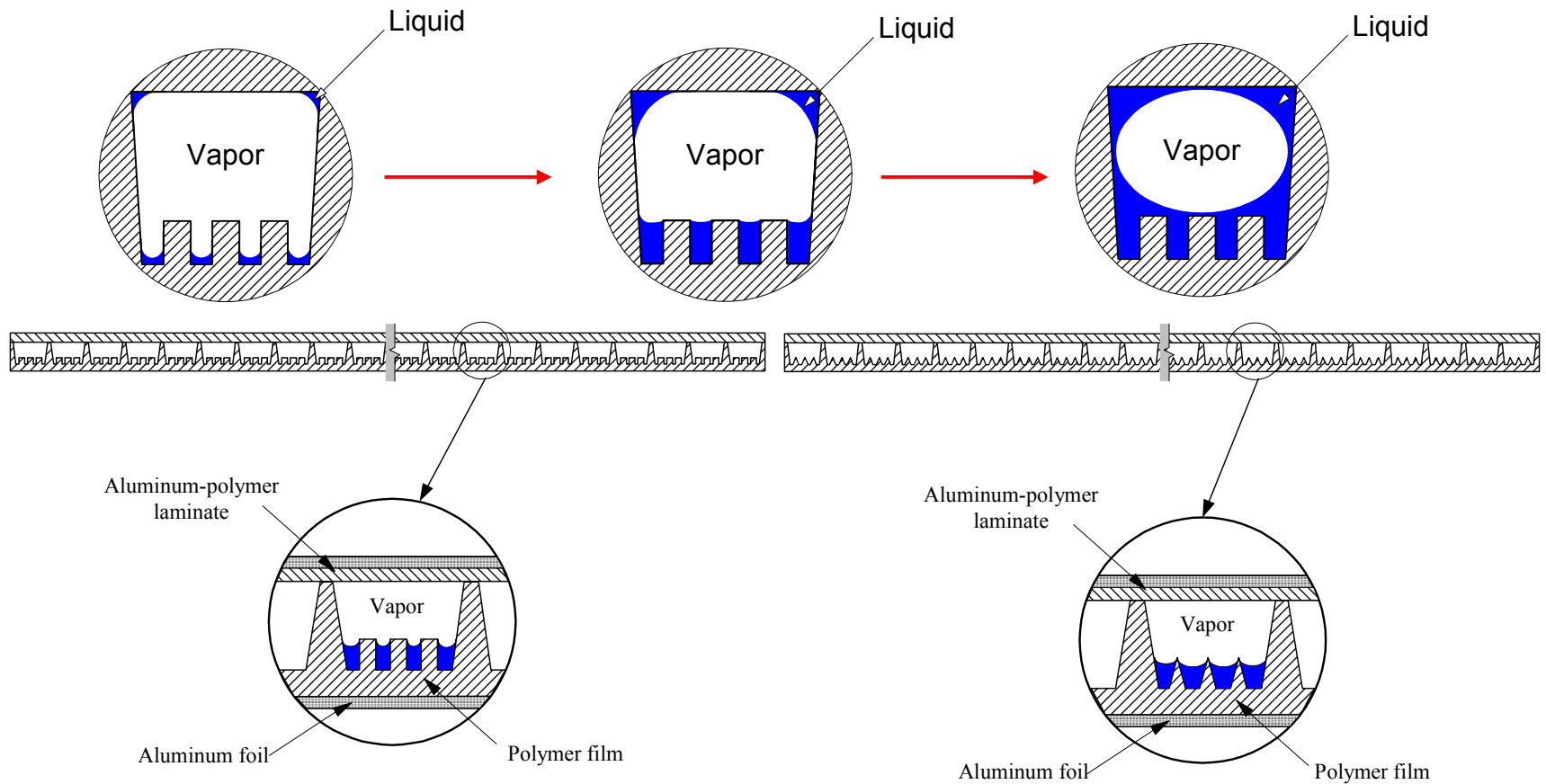
Capillary Limit - Methanol



Capillary Limit Water



Stable Liquid Configurations



Summary and Conclusions

- Two new micro heat pipe concepts have been developed
- Wire Bonded heat pipe arrays with an effective conductivity of 30 times that of solid aluminum have been developed and tested.
- Flexible polymer heat pipes have been fabricated and modeled.
- These polymer heat pipes offer a greater degree of flexibility and a potentially higher effective thermal conductivity than any previously developed.
- Applications of the these two concepts have a wide range of applications that extends well beyond spacecraft radiators.